

Abstract

Annular combustion chambers (4) for a gas turbine (1) and gas turbine (1)

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The invention relates to an annular combustion chamber (4) for a gas turbine (1) wherein the annular combustion chamber (4) extends in an axial direction (A), encloses a combustor (7), and has on its inside facing the combustor (7) a bearing structure (26) on which a lining element (10) secured to this lines the annular combustion chamber (4). The object is to disclose an annular combustion chamber (4) with a lining element (10) that meets the mechanical requirements while at the same time taking account of the system's maintenance-friendliness. The object is achieved in that the annular combustion chamber (4) has a lining element (10) wherein (10) on the rear side (13) facing away from the combustor (7) of two edge areas (15) on the lining element a plurality of interlocking means (11) are located which have a hook width (B), and wherein the lining element (10) is secured to the corresponding bearing structure (26) such that in order to release the lining element (10) from the bearing structure (26) the lining element (10) is moved by the extent of the hook width (B) of the interlocking means (11) in the axial direction (A).

25 Fig. 3